

HPC approaches in industry

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- Pipistrel
- Motivation
- Fortissimo project
- Mikelangelo project





Green Flight Challenge
2011

Sponsored by
Google

Prizes provided by
NASA

Conducted by
CAFE FOUNDATION

Taurus G4



Panthera

Motivation



- Pipistrel cluster: 2 x (8 cores, 66GB RAM)
- Typical simulations:
 - fully turbulent RANS simulations
 - low-Re airfoil simulations
 - 5M (15M max) cells mesh



Motivation









- Pipistrel cluster: 2 x (8 cores, 66GB RAM)
- Typical simulations:
 - fully turbulent RANS simulations
 - low-Re airfoil simulations
 - 5M (15M max) cells mesh
- Pipistrel needs on-demand resources to increase fidelity of the airflow simulations
- Requirements:
 - Resources cheap to use
 - Fast to spin-up
 - Self-service
 - Remote visualization
 - OpenFOAM support
 - Supporting different modes: MPI and EP



- Experiment: Cloud-based simulation of aerodynamics of light aircraft
- Partners: End User: PIPISTREL
HPC Expert: XLAB
HPC Provider: ARCTUR
- Application: OpenFOAM 2.2.0
- www.fortissimo-project.eu/



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- laminar-turbulent transition modelling
with RANS simulations: low-Re $k - k_L - \omega$ turbulence model

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- laminar-turbulent transition modelling
with RANS simulations: low-Re $k - k_L - \omega$ turbulence model
- complete Panthera aircraft at cruise speed ($Re=5.7e6$)

Course of action:

- Simple test cases with turbulent model $k - k_L - \omega$
- A wing at smaller velocities
- A wing at cruise velocity
- Complete Panthera aircraft at smaller velocities
- Complete Panthera aircraft at cruise speed

Course of action:

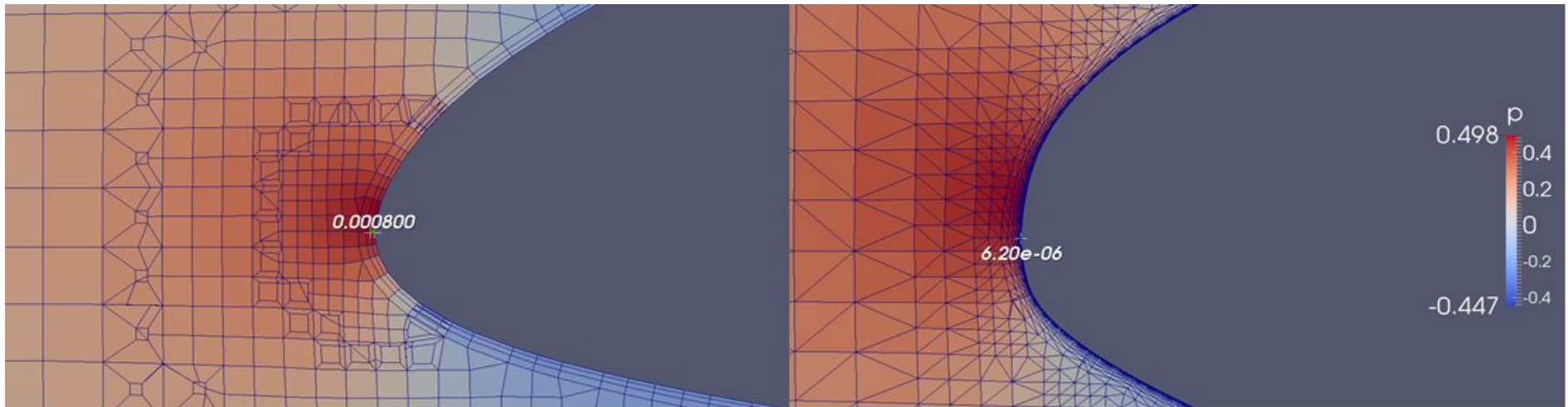
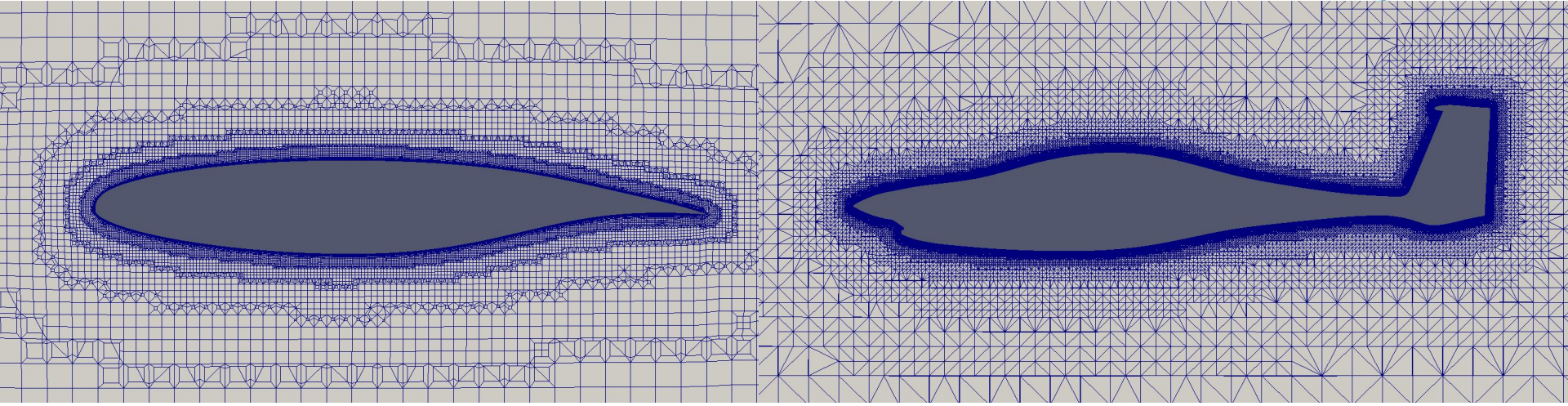
- Simple test cases with turbulent model $k - k_L - \omega$
- A wing at smaller velocities
- A wing at cruise velocity
- Complete Panthera aircraft at smaller velocities
- Complete Panthera aircraft at cruise speed

	<u>In house cluster</u>	<u>Arctur's HPC</u>
mesh size	5 -10M cells	115M cells
thinnest layer	~ 0.1mm	~ 0.006mm
No. cores	8	60 - mesh 180 - simul.
simulation time	1-2 days	2-3 days

Mesh



FORTISSIMO



June 30, 2016

IHPCSS2016, Ljubljana

snappyHexMeshDict:

```
addLayersControls
{
    relativeSizes false;
    layers
    {
        "(body).*"
        {
            nSurfaceLayers 13;
        }
    }
    expansionRatio 1.5;
    finalLayerThickness 0.0008;

    featureAngle 30;
    slipFeatureAngle 0;
}
```

```
meshQualityControls
{
    maxNonOrtho 65;

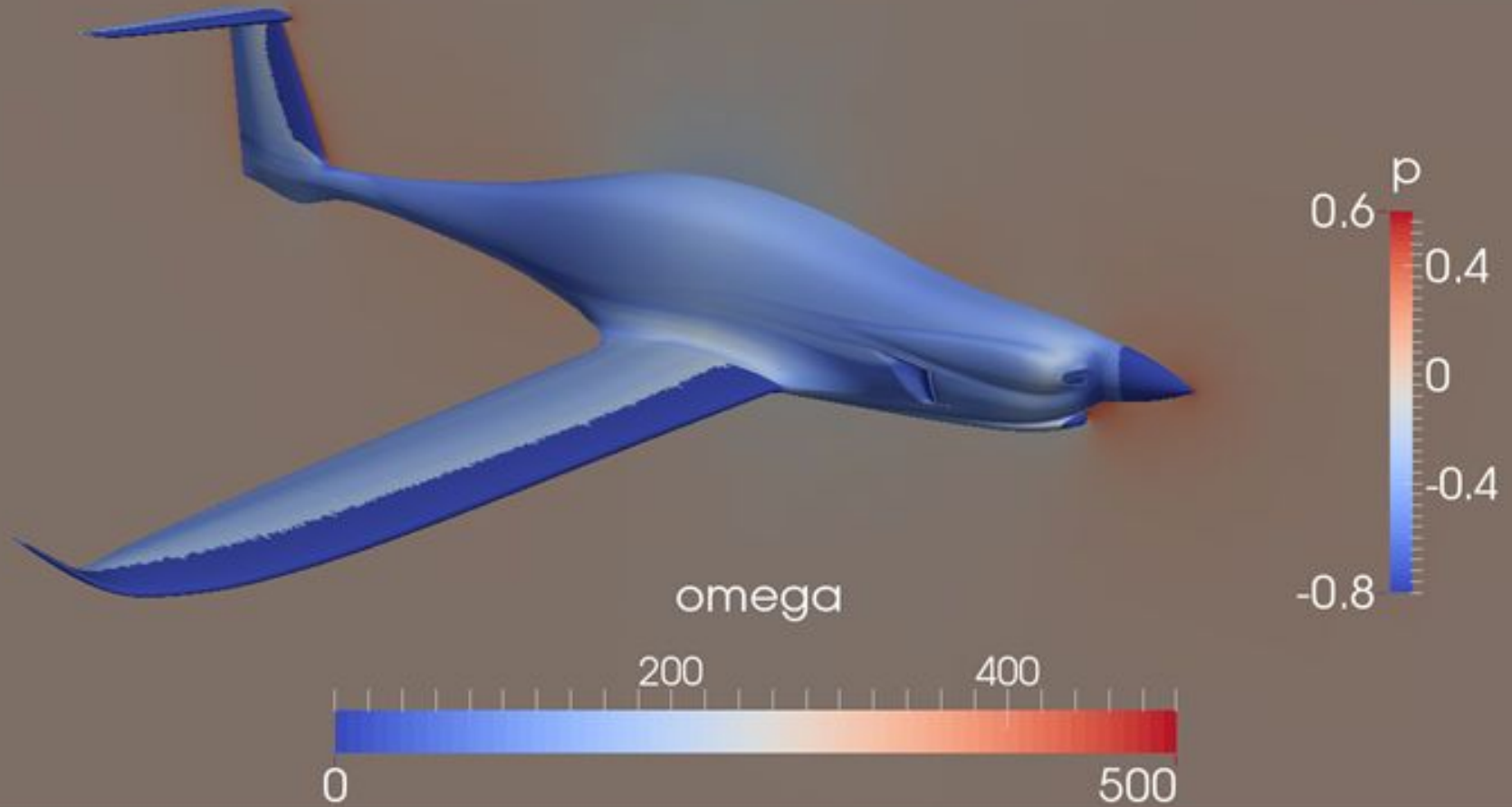
    maxBoundarySkewness -20;
    maxInternalSkewness -4;

    minDeterminant 1e-6;
}
```

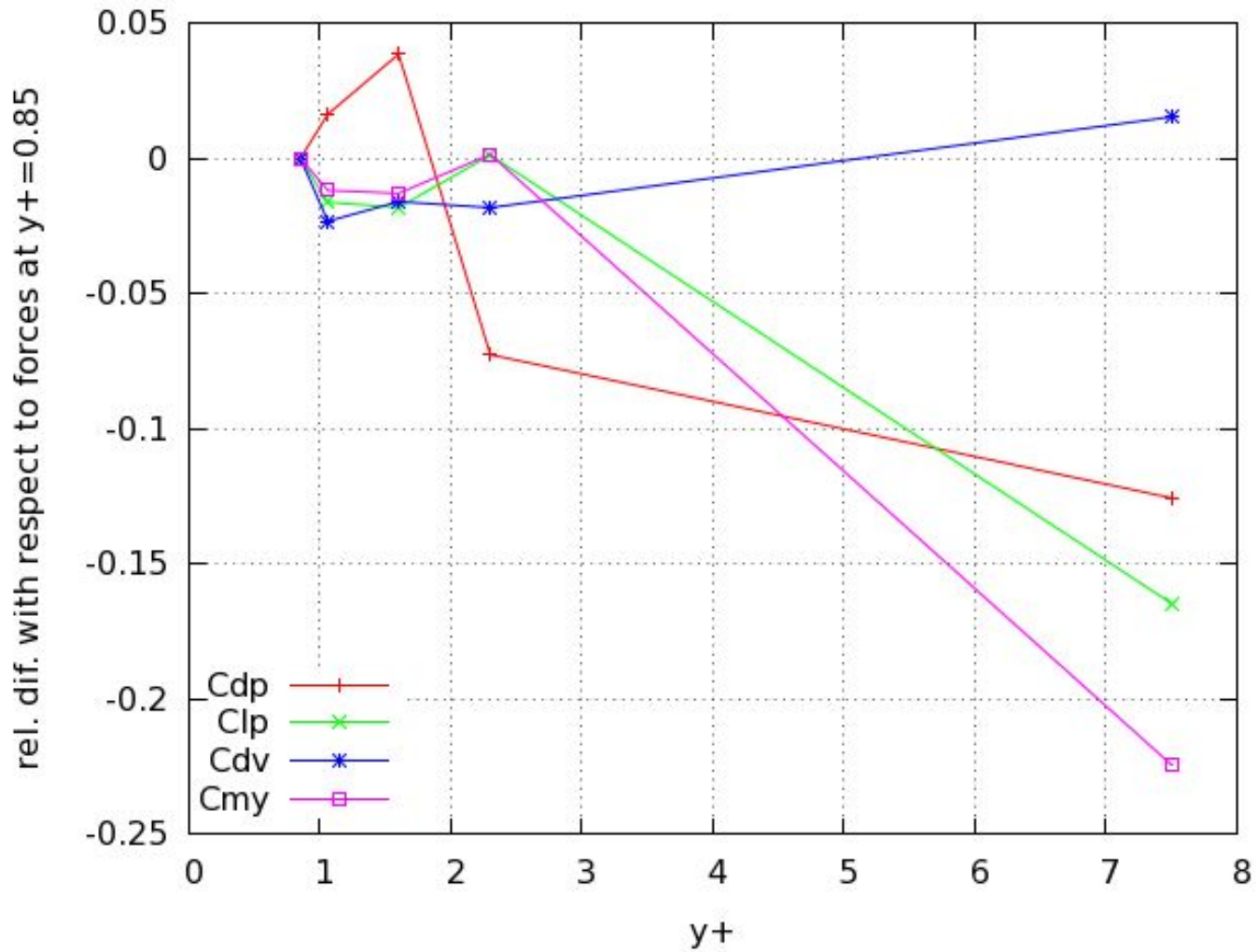
Results



FORTISSIMO



Results



Learn how to:

- make a proper mesh - such a fine mesh at the surface
- use symmetry plane
- preview the decomposed case - reconstruction takes a lot of time
- extract only necessary data and preview it with paraView
- automatically consecutively run all steps of the simulation process
- how to run, handle and postprocess such big cases
- persuade HPC provider to increase RAM

Mikelangelo



ONE technology stack
UNIFIED for HPC and Cloud

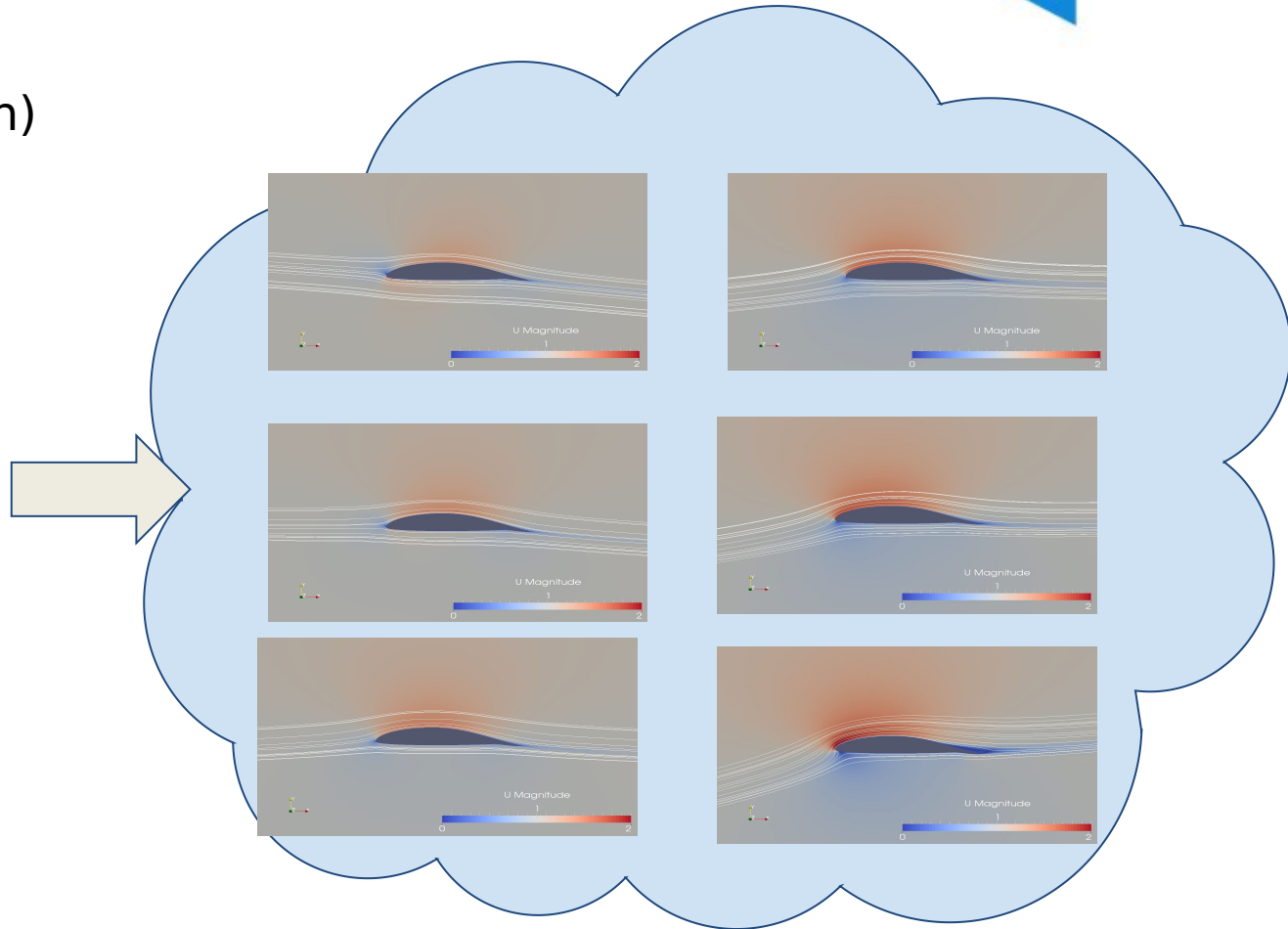
MISSION: To provide cloud community
with technologies for

- fast, agile and secure Cloud and HPC application deployments in
- diverse hardware environments,
- through packaged applications,
- using lean guest operating system OSv
- superfast hypervisor – sKVM

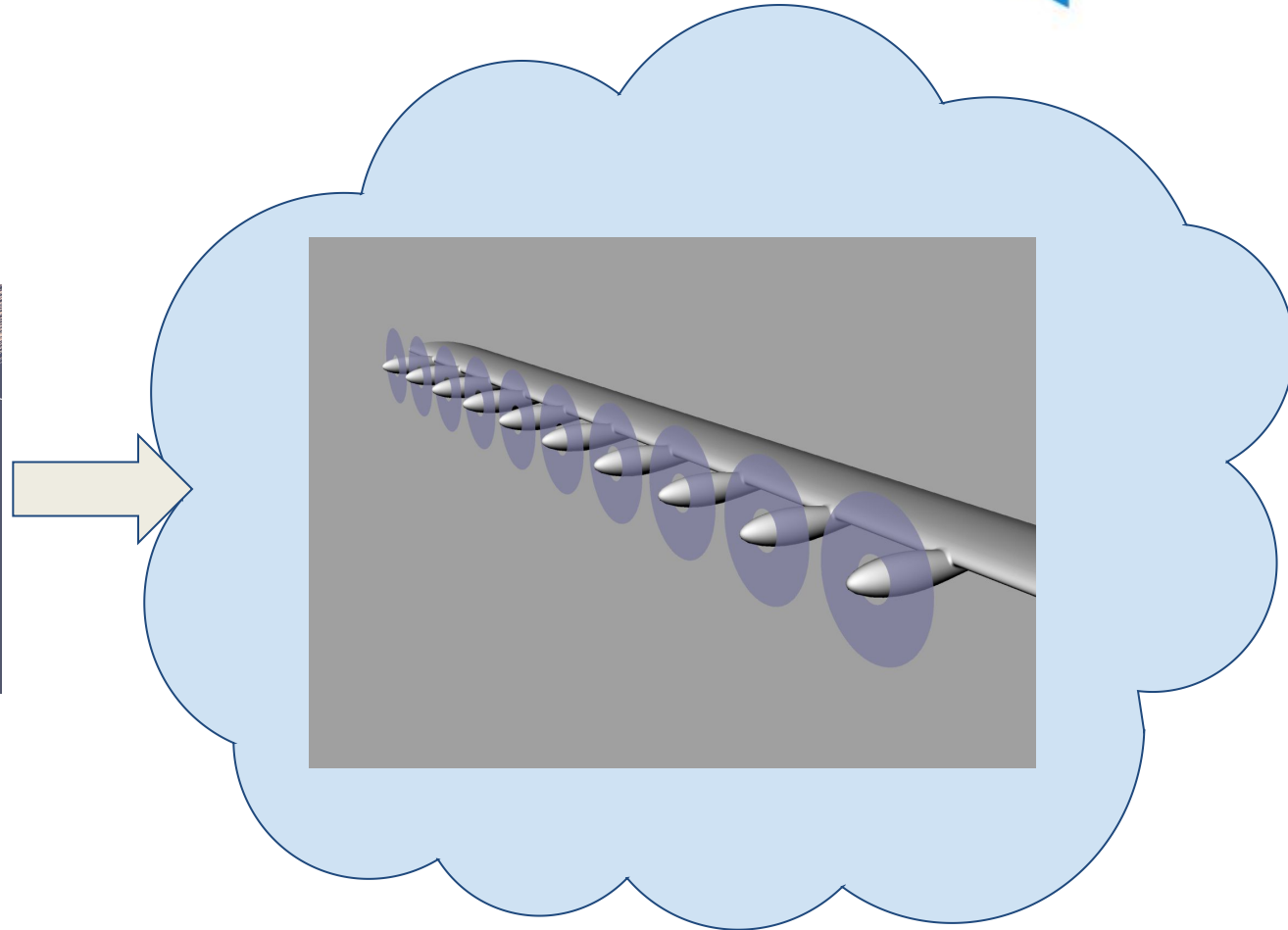
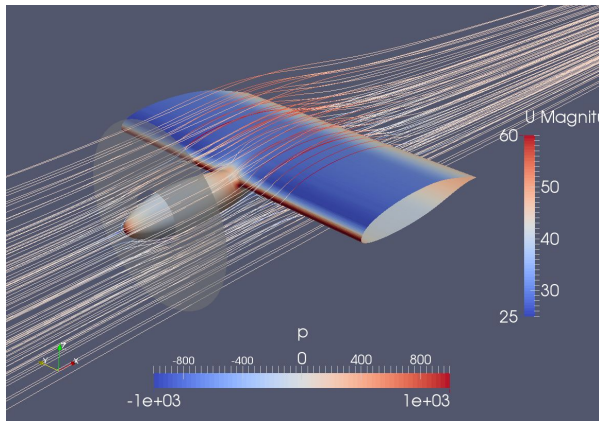


www.mikelangelo-project.eu

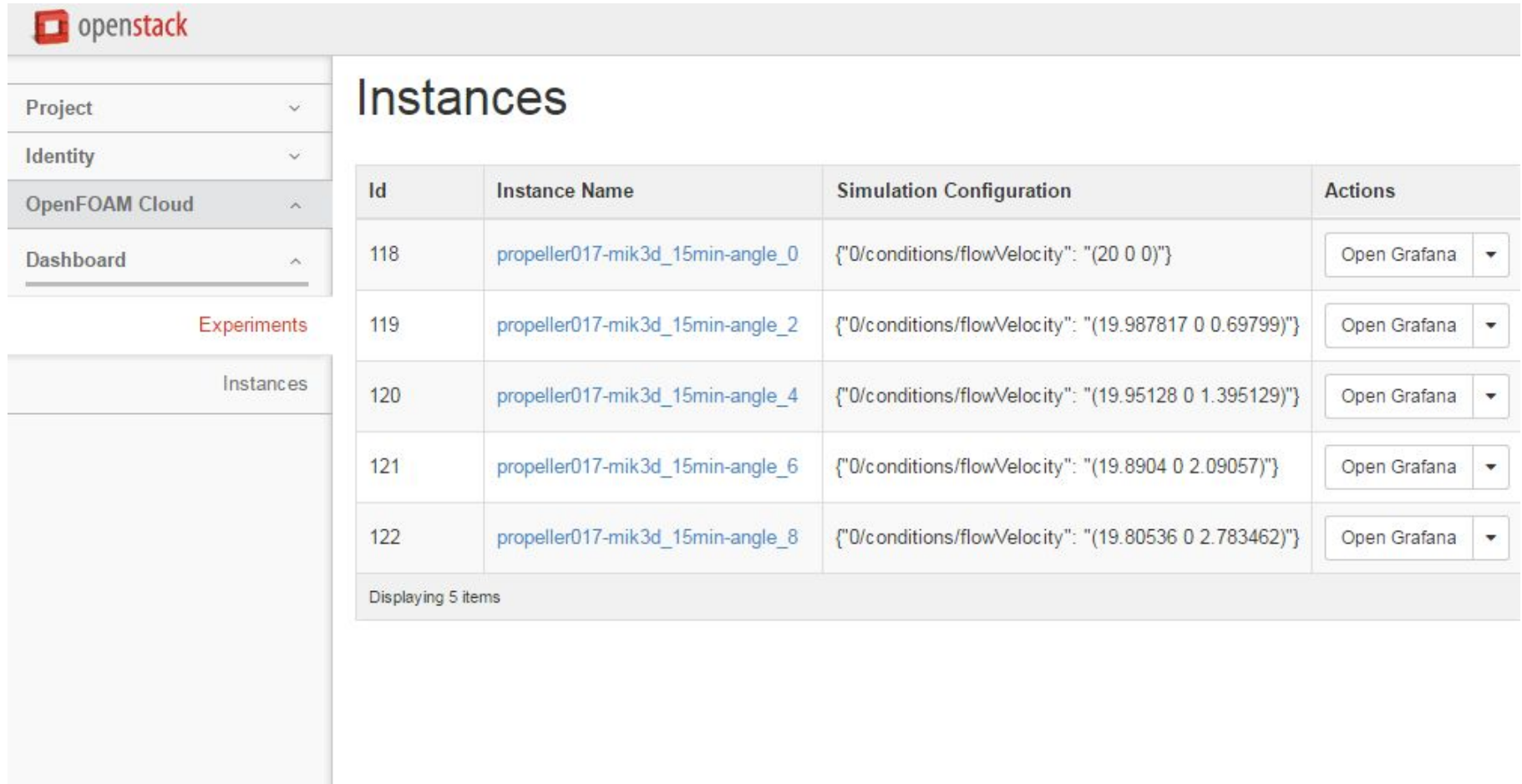
EP use-case
(parameter exploration)



MPI use-case
(typical HPC use case)



OpenFOAM Cloud application




The screenshot shows the OpenStack dashboard interface. On the left, there is a navigation menu with the following items: Project, Identity, OpenFOAM Cloud, and Dashboard. Below the menu, there are two sections: 'Experiments' and 'Instances'. The 'Instances' section is active, displaying a table of instances.

Id	Instance Name	Simulation Configuration	Actions
118	propeller017-mik3d_15min-angle_0	<code>{"0/conditions/flowVelocity": "(20 0 0)"}</code>	Open Grafana ▼
119	propeller017-mik3d_15min-angle_2	<code>{"0/conditions/flowVelocity": "(19.987817 0 0.69799)"}</code>	Open Grafana ▼
120	propeller017-mik3d_15min-angle_4	<code>{"0/conditions/flowVelocity": "(19.95128 0 1.395129)"}</code>	Open Grafana ▼
121	propeller017-mik3d_15min-angle_6	<code>{"0/conditions/flowVelocity": "(19.8904 0 2.09057)"}</code>	Open Grafana ▼
122	propeller017-mik3d_15min-angle_8	<code>{"0/conditions/flowVelocity": "(19.80536 0 2.783462)"}</code>	Open Grafana ▼

Displaying 5 items

OpenFOAM Cloud application



Project
Identity
OpenFOAM Cloud
Dashboard
Experiments
Instances

Instance Details: propeller017-mik3d_15min-angle_8

Open Grafana

Overview Configurations Log

Log Length 35 Go

Instance Console Log

```
Time = 498
smoothSolver: Solving for Ux, Initial residual = 8.59654183e-09, Final residual = 8.59654183e-09, No Iterations 0
smoothSolver: Solving for Uy, Initial residual = 3.40398342e-07, Final residual = 2.91711568e-08, No Iterations 4
smoothSolver: Solving for Uz, Initial residual = 4.01008674e-08, Final residual = 6.02654984e-09, No Iterations 3
GANG: Solving for p, Initial residual = 1.02734664e-07, Final residual = 1.17029414e-08, No Iterations 1
time step continuity errors : sum local = 1.13807502e-08, global = 7.52746917e-11, cumulative = 0.000231720351
smoothSolver: Solving for nuTilda, Initial residual = 1.98934906e-07, Final residual = 1.12326154e-08, No Iterations 3
ExecutionTime = 1011.20418 s  ClockTime = 1011 s

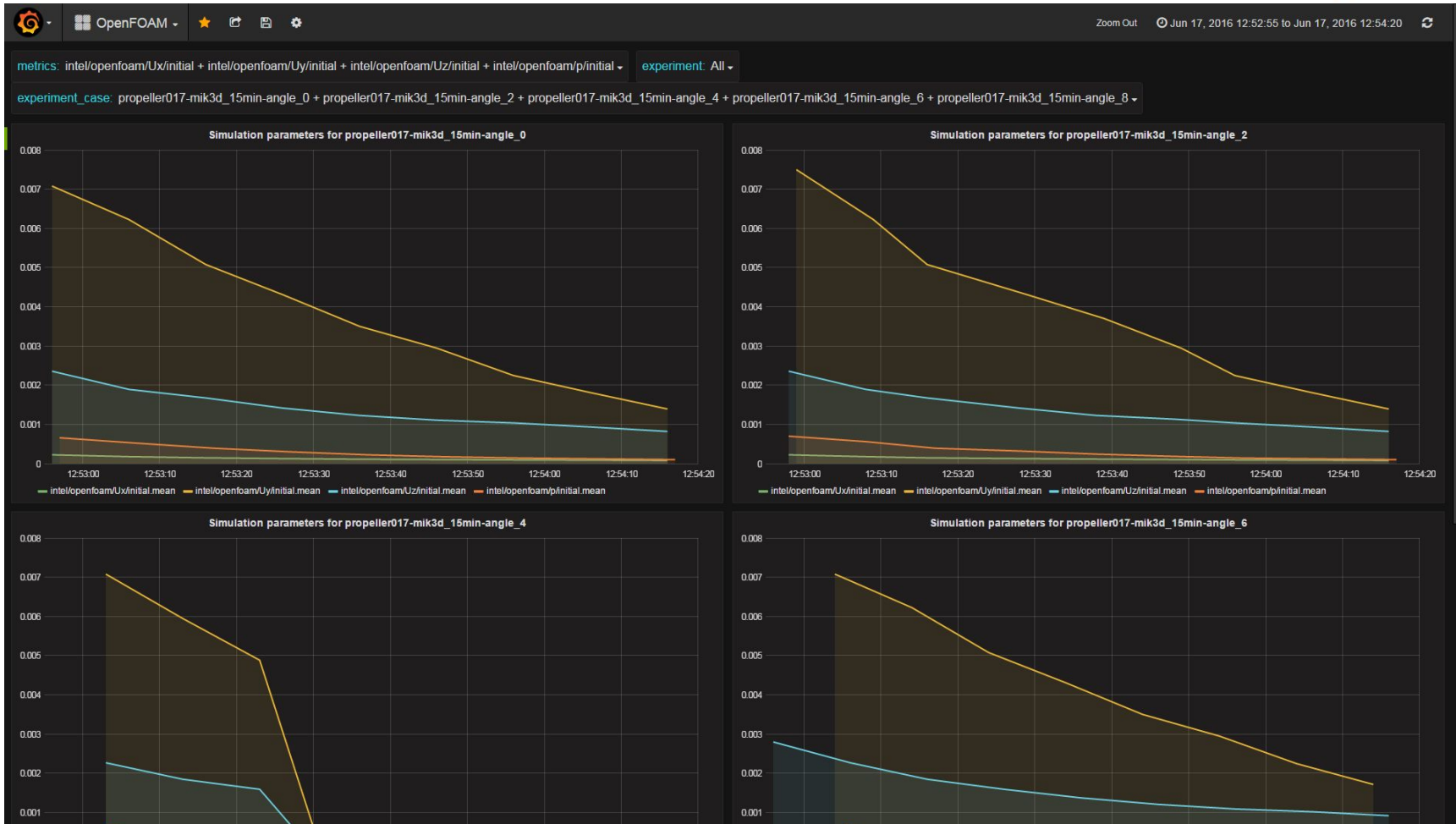
Time = 499
smoothSolver: Solving for Ux, Initial residual = 6.59538424e-09, Final residual = 6.59538424e-09, No Iterations 0
smoothSolver: Solving for Uy, Initial residual = 4.06215896e-07, Final residual = 3.54231039e-08, No Iterations 4
smoothSolver: Solving for Uz, Initial residual = 4.82948477e-08, Final residual = 7.20537599e-09, No Iterations 3
GANG: Solving for p, Initial residual = 8.92147218e-08, Final residual = 8.92147218e-08, No Iterations 0
time step continuity errors : sum local = 8.6758571e-08, global = 7.05776354e-11, cumulative = 0.000231720421
smoothSolver: Solving for nuTilda, Initial residual = 2.15646728e-07, Final residual = 2.14679309e-08, No Iterations 2
ExecutionTime = 1012.5037 s  ClockTime = 1012 s

Time = 500
smoothSolver: Solving for Ux, Initial residual = 8.47967314e-09, Final residual = 8.47967314e-09, No Iterations 0
smoothSolver: Solving for Uy, Initial residual = 3.36074288e-07, Final residual = 2.94551483e-08, No Iterations 4
smoothSolver: Solving for Uz, Initial residual = 3.89361536e-08, Final residual = 5.86469592e-09, No Iterations 3
GANG: Solving for p, Initial residual = 9.95062099e-08, Final residual = 9.95062099e-08, No Iterations 0
time step continuity errors : sum local = 9.6766726e-08, global = 8.60675765e-11, cumulative = 0.000231720507
smoothSolver: Solving for nuTilda, Initial residual = 1.99244634e-07, Final residual = 1.13033085e-08, No Iterations 3
ExecutionTime = 1021.2018 s  ClockTime = 1021 s

forces forceCoeffs1 output:
sum of forces:
  pressure : (3.25168492 0.000271958824 47.0743495)
  viscous   : (0.812352192 1.52393965e-05 0.0175473364)
  porous    : (0 0 0)
sum of moments:
  pressure : (-0.0032130266 23.650434 -0.00123414582)
  viscous   : (-6.74220369e-06 0.0259565388 -7.47318381e-06)
  porous    : (0 0 0)

End
```

OpenFOAM Cloud application



Thank you for your attention!

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